

1. Air flows through a nozzle of diameter 1.3 mm having a discharge coefficient of 0.95, from a pressure of 5 atm to a pressure of 1 atm at 25 °C. Density of air will be:

- A) 5.930 g/cm<sup>3</sup>
- B) 59.30 kg/cm<sup>3</sup>
- C) 0.593 kg/m<sup>3</sup>
- D) None

Ans (A)

$$\rho = \frac{pM}{RT} = \frac{5 \times 101325 \times 29}{8314 \times 298} = 5.93 \text{ kg m}^{-3}$$

2. For Q. 1 Maximum velocity will be:

- A) 3.15 m/s
- B) 2.95 m/s
- C) 0.315 km/s
- D) 3150 m/s

Answer (C)

Now, upstream pressure is 5 times greater than that of downstream pressure. Hence pressure ratio is at critical condition

$$v_0 = \sqrt{\frac{2\gamma p}{(\gamma - 1)\rho} \left[ 1 - \left( \frac{p_0}{p} \right)_{cr}^{\frac{\gamma-1}{\gamma}} \right]}$$

$$= \sqrt{\frac{2 \times 1.4 \times 5 \times 101325}{(1.4 - 1) \times 5.93} \left[ 1 - (0.528)^{\frac{1.4-1}{1.4}} \right]}$$

$$= 315.83 \text{ ms}^{-1}$$

$$= 0.315 \text{ kms}^{-1}$$

3. For Q. 1 Mass flow rate will be:

- A) 669 × 10<sup>-6</sup> kg/s
- B) 24.09 Kg/h
- C) Both A&B
- D) None of these

Ans. (A)

$$\text{Now, } A_0 = \frac{\pi}{4} (0.0013)^2 = 1.327 \times 10^{-6} \text{ m}^2$$

$$\begin{aligned} \therefore W &= C_D A_0 \sqrt{\frac{2\gamma P \rho}{(\gamma - 1)} \left[ \left( \frac{P_0}{P} \right)_{cr}^{\frac{2}{\gamma}} - \left( \frac{P_0}{P} \right)_{cr}^{\frac{\gamma+1}{\gamma}} \right]} \\ &= 0.95 \times 1.327 \times 10^{-6} \times \sqrt{\frac{2 \times 1.4 \times 101325 \times 5.93}{(1.4 - 1)} \left[ (0.528)^{\frac{2}{1.4}} - (0.528)^{\frac{1.4+1}{1.4}} \right]} \\ &= 6.69 \times 10^{-4} \text{ kg s}^{-1}, \text{ or, } 2.409 \text{ kg h}^{-1} \end{aligned}$$

4. Methane is being pumped through a 50.0 cm ID pipeline for a distance of  $1.0 \times 10^5$  m at a rate of 2.0 Kg Mole/s. Mass density of methane is

- 40.746 kg/m<sup>2</sup>.s
- 162.984 kg/m<sup>2</sup>.s
- 1273.31 kg/m<sup>2</sup>.s
- 5093.25 kg/m<sup>2</sup>.s

Ans. (b)

$$G = 2.0 \frac{\text{Kg Mole}}{\text{s}} \times 16 \frac{\text{Kg}}{\text{Kg Mole}} \times \frac{1}{\frac{\pi D^2}{4} \text{m}^2} = \frac{2 \times 16 \times 4}{3.1416 \times 5^2} = 162.984 \text{ kg/m}^2.\text{s}$$

5. For adiabatic flow,  $PV^\gamma = C$  where  $\gamma$  is

- $\gamma$  is the ratio of heat capacities at constant pressure and at constant volume respectively.
- $\gamma$  is the ratio of heat capacities at constant volume and at constant pressure respectively.
- $\gamma$  is always 1
- None

Ans. (a)

$$\gamma = C_p/C_v$$

6. Discharge through nozzle is maximum when,

- $P/P_0 = 0.528$
- $P_0/P = 0.528$
- $P/P_0 < 0.528$
- All of the above

Ans. (b)

7. A nozzle of 1 mm dia with a coefficient of discharge of 0.92 is to deliver air from 4 atm pressure to 3 atm pressure at 35 °C. Density of air:

- 4.59 kg/m<sup>3</sup>
- 3.44 kg/m<sup>3</sup>

- c) 4.59 g/m<sup>3</sup>
- d) 3.44 g/m<sup>3</sup>

Ans. (a)

$$\rho = \frac{pM}{RT} = \frac{4 \times 101325 \times 28.97}{8314 \times 308} = 4.59 \text{ kg/m}^3$$

8. In Q. 8 Velocity of air

- a) 260.15 m/s
- b) 220.85 m/s
- c) 220.85 cm/s
- d) None

Ans. (b)

$$\text{We know that, } v_0 = \sqrt{\frac{2\gamma p}{(\gamma - 1)\rho} \left[ 1 - \left( \frac{p_0}{p} \right)^{\frac{\gamma-1}{\gamma}} \right]}$$

$$v_0 = \sqrt{\frac{2 \times 1.4 \times 4 \times 101325}{(1.4 - 1) \times 4.59} \left[ 1 - \left( \frac{3}{4} \right)^{\frac{1.4-1}{1.4}} \right]}$$

$$= 220.85 \text{ m/s}$$

9. In Q. 8 Mass flow rate

- a) 2.48 kg/h
- b) 2.15 kg/h
- c) 2.48 kg/s
- d) 2.15 kg/s

Ans. (b)

$$A_0 = \frac{\pi D^2}{4} = \frac{\pi \times 0.01^2}{4} = 7.85 \times 10^{-7} \text{ m}^2$$

Now, we know that

$$W = C_D A_0 \sqrt{\frac{2\gamma p \rho}{(\gamma - 1)} \left[ \left( \frac{p_0}{p} \right)^{\frac{2}{\gamma}} - \left( \frac{p_0}{p} \right)^{\frac{\gamma+1}{\gamma}} \right]}$$

$$W = 0.92 \times 7.85 \times 10^{-7} \sqrt{\frac{2 \times 1.4 \times 4 \times 101325 \times 4.59}{(1.4 - 1)} \left[ \left( \frac{3}{4} \right)^{2/1.4} - \left( \frac{3}{4} \right)^{\frac{1.4+1}{1.4}} \right]}$$

$$= 0.000597 \text{ Kg/s} = 2.15 \text{ Kg/h}$$

10. In Q. 8 Maximum velocity of air

- a) 370.89 m/s
- b) 321.20 m/s
- c) 260.15 m/s
- d) 220.85 m/s

Ans. (a)

At critical pressure ratio, i.e.,  $p_0/p = 0.528$ , the velocity will be maximum

$$v_0 = \sqrt{\frac{2\gamma p}{(\gamma - 1)\rho} \left[ 1 - \left( \frac{p_0}{p} \right)_{cr}^{\frac{\gamma-1}{\gamma}} \right]}$$

11. In Q. 8 Maximum mass flow rate is

- a) 2.594 kg/h
- b) 3 kg/h
- c) 2.48 kg/h
- d) None

Ans. (b)

At critical pressure ratio, i.e.,  $p_0/p = 0.528$ , the discharge will be maximum

12. The unit of rate of mass discharge from the nozzle “W” will be

- a) kg/hr

- b) kg-hr
- c) kg-m/s
- d) None of the above

Ans. (a)

13. The expression for the Bernoulli's equation is

- a)  $\frac{p}{\rho} + \frac{v^2}{2} + gz = \text{constant}$
- b)  $\frac{p}{\rho} + \frac{v^2}{2g} + gz = \text{constant}$
- c)  $\frac{p}{\rho g} + \frac{v^2}{2g} + gz = \text{constant}$
- d)  $\frac{p}{\rho} + \frac{v^2}{2g} + z = \text{constant}$

Ans. (a)

14. For adiabatic flow

- a)  $pV^\gamma = C$
- b)  $pV = C$
- c)  $pV^{\gamma-1} = C$
- d) None of the above

Ans. (a)